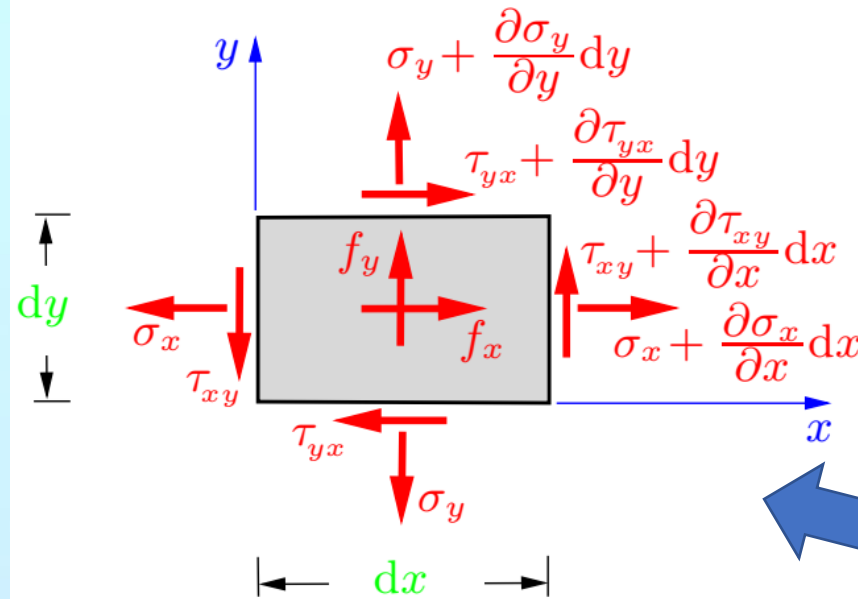


# 3 Strain, Hooke's Law

## 3.1 State of Strain

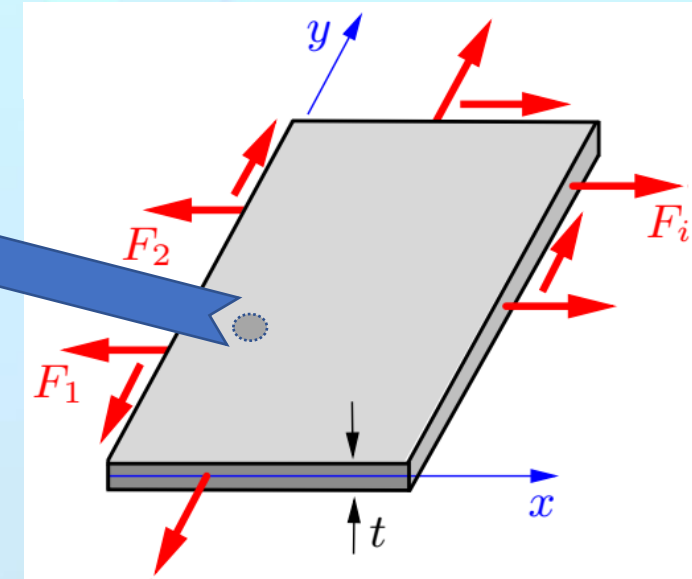
## 3.2 Hooke's Law

Two Dimensional case



$$\boldsymbol{\sigma} = \begin{bmatrix} \sigma_x & \tau_{xy} \\ \tau_{xy} & \sigma_y \end{bmatrix}$$

$\sigma_x(x, y), \sigma_y(x, y) & \tau_{xy}(x, y)??$



$$\frac{\partial \sigma_x}{\partial x} + \frac{\partial \tau_{xy}}{\partial y} + f_x = 0$$

$$\frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \sigma_y}{\partial y} + f_y = 0$$

### 3.1 State of Strain

The uniaxial deformation of a tension bar, was described in first lecture by two kinematic quantities: displacement  $u$  and strain  $\epsilon = du/dx$ .

The length of the elongated element is:  $dx + (u + du) - u = dx + du$ .

The longitudinal strain is: 
$$\epsilon(x) = \frac{dx + du - dx}{dx} = \frac{du}{dx}$$

Here the deformation of a plane structure will be described

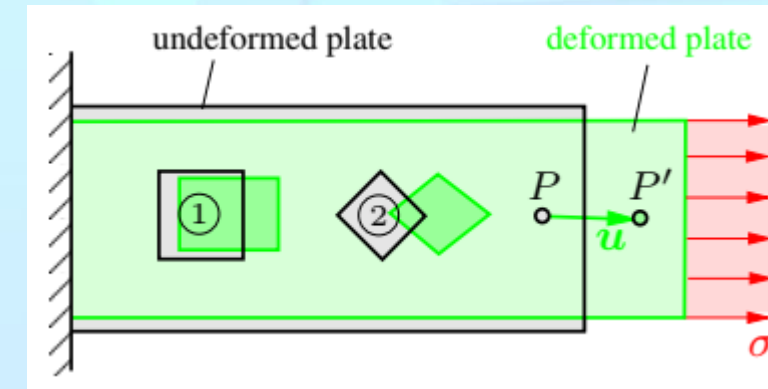
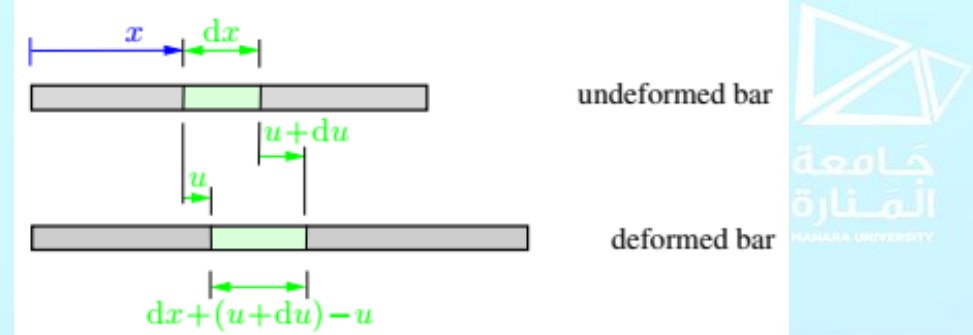
To this end a panel or a sheet is considered where two squares ① and ② are marked such that they are tilted against each other.

When the panel is loaded by a normal stress  $\sigma$ , point  $P$  experiences a displacement  $\mathbf{u} \equiv \vec{u}$  from its initial position to the new position  $P'$ .

Since the *displacement vector*  $\vec{u}$  depends on the location, the side lengths (square ①) and the side lengths and angles (square ②), respectively, are changed during the deformation.

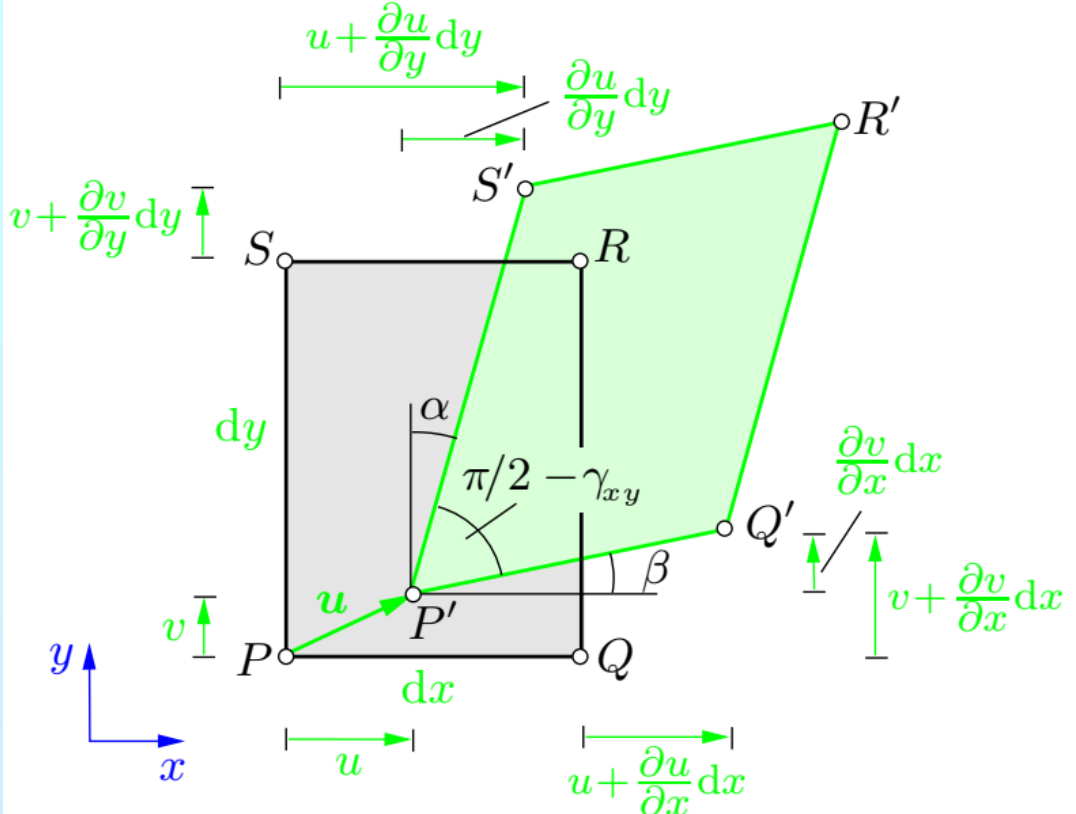
In the discussions that follow the changes of side lengths and angles are considered under the assumption of *small* deformations: Changes in lengths and angles are very small.

$$\vec{u}(x, y) = u(x, y)\vec{i} + v(x, y)\vec{j}, \quad \frac{\partial u}{\partial x}, \frac{\partial u}{\partial y}, \frac{\partial v}{\partial x} \text{ and } \frac{\partial v}{\partial y} \text{ are all much smaller than 1.}$$



Next figure shows an infinitesimal rectangle  $PQRS$  of side lengths  $dx$  and  $dy$  in the undeformed state. During the deformation it is transformed into the new position  $P'Q'R'S'$ .

Displacement  $\vec{u}(x, y)$  of  $P(x, y)$  has components:  $u(x, y)$  in  $x$  direction and  $v(x, y)$  in  $y$  direction. displacements of points, adjacent to  $P$ , can be described with the help of Taylor expansions. For the functions  $u$  and  $v$ , depending on  $x$  and  $y$ , as:



$$\left. \begin{aligned}
 u(x + dx, y + dy) &= u(x, y) + \frac{\partial u(x, y)}{\partial x} dx + \frac{\partial u(x, y)}{\partial y} dy + \dots \\
 v(x + dx, y + dy) &= v(x, y) + \frac{\partial v(x, y)}{\partial x} dx + \frac{\partial v(x, y)}{\partial y} dy + \dots
 \end{aligned} \right\} \text{Simplified for } Q(dy = 0) \text{ \& } S(dx = 0).$$

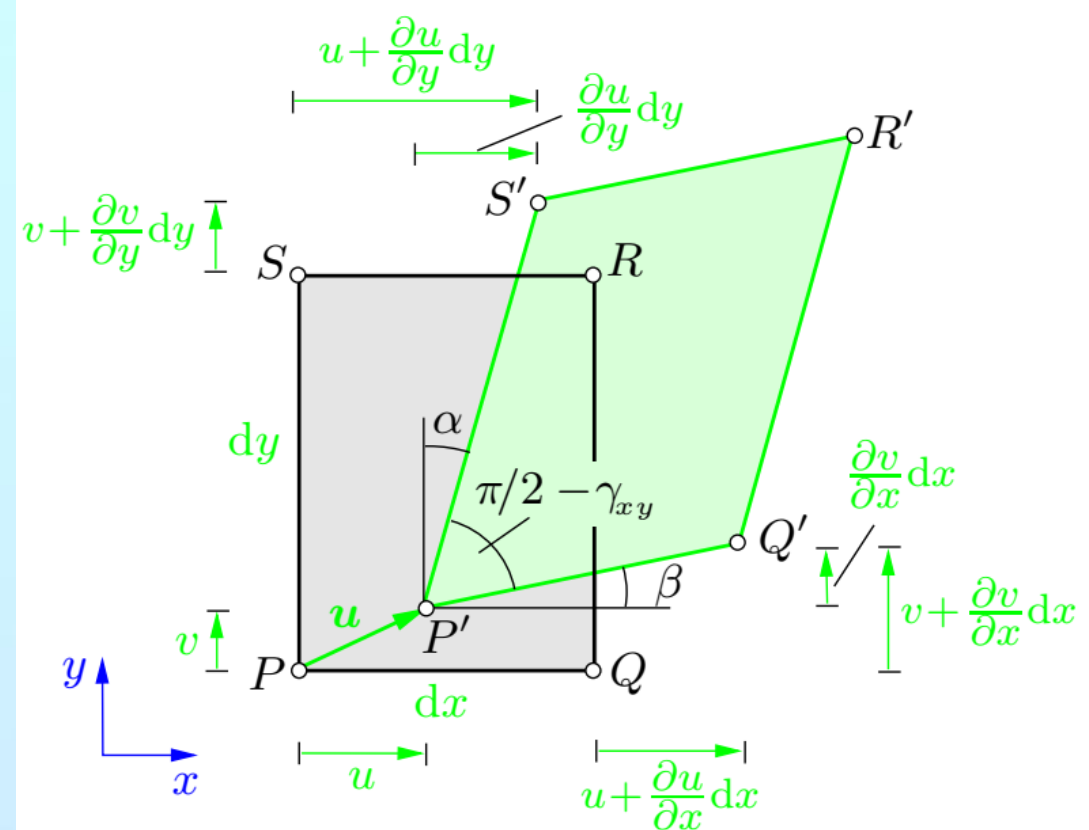
For point  $Q(dy = 0)$ :  $u(x + dx, y) = u(x, y) + \frac{\partial u(x, y)}{\partial x} dx$  &  $v(x + dx, y) = v(x, y) + \frac{\partial v(x, y)}{\partial x} dx$

For point  $S(dx = 0)$ :  $u(x, y + dy) = u(x, y) + \frac{\partial u(x, y)}{\partial y} dy$  &  $v(x, y + dy) = v(x, y) + \frac{\partial v(x, y)}{\partial y} dy$

For point  $Q$ :  $u(x + dx, y) = u(x, y) + \frac{\partial u(x, y)}{\partial x} dx$   
 $v(x + dx, y) = v(x, y) + \frac{\partial v(x, y)}{\partial x} dx$

For point  $S$ :  $u(x, y + dy) = u(x, y) + \frac{\partial u(x, y)}{\partial y} dy$   
 $v(x, y + dy) = v(x, y) + \frac{\partial v(x, y)}{\partial y} dy$

During deformation, line  $PQ$  is transformed into line  $P'Q'$ . Since small deformations are assumed ( $\beta \ll 1, \cos \beta \approx 1$ ), the lengths of  $P'Q'$  and of its projection on  $x$  are approximately the same. So



$$|\overrightarrow{P'Q'}| \approx \left( x + dx + u + \frac{\partial u}{\partial x} dx \right) - (x + u) = dx + \frac{\partial u}{\partial x} dx$$

Then the *normal strain*  $\epsilon_x$  in the  $x$ -direction defined as the ratio of length increment to initial length, is

$$\epsilon_x = \frac{|\overrightarrow{P'Q'}| - |\overrightarrow{PQ}|}{|\overrightarrow{PQ}|} = \frac{dx + \frac{\partial u}{\partial x} dx - dx}{dx} \Rightarrow \epsilon_x = \frac{\partial u}{\partial x}$$

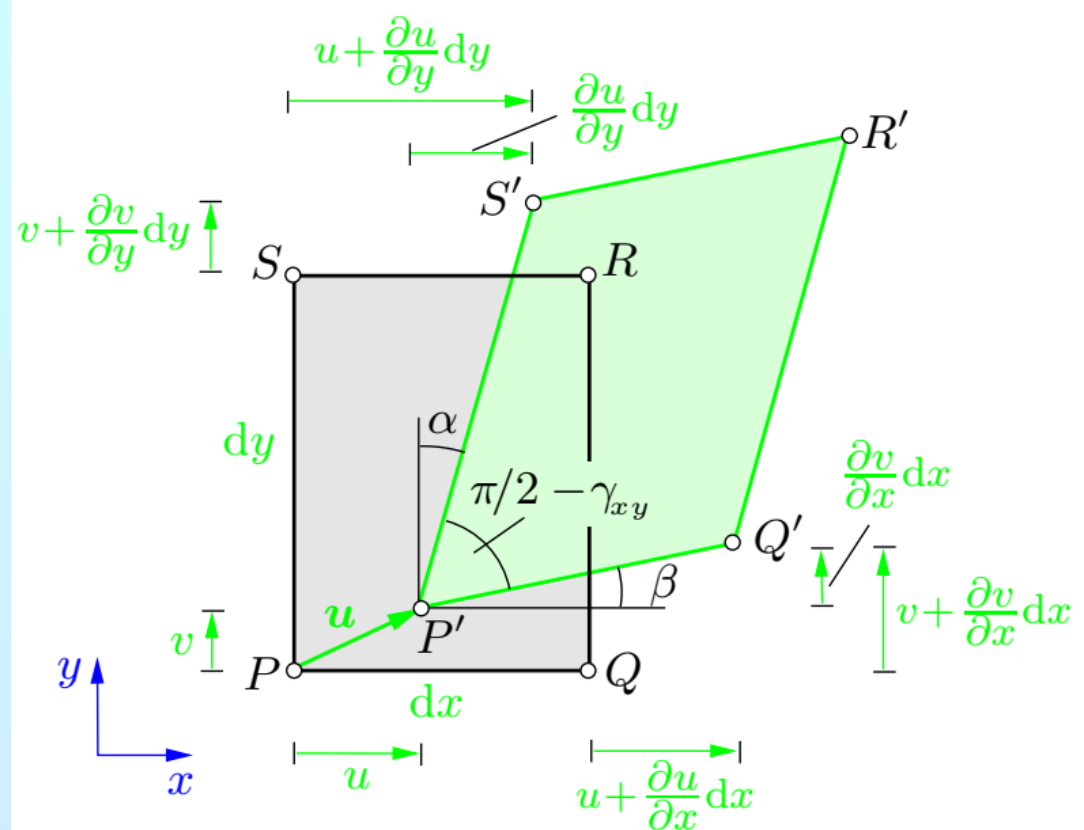
For point  $Q$ :  $u(x + dx, y) = u(x, y) + \frac{\partial u(x, y)}{\partial x} dx$

$v(x + dx, y) = v(x, y) + \frac{\partial v(x, y)}{\partial x} dx$

For point  $S$ :  $u(x, y + dy) = u(x, y) + \frac{\partial u(x, y)}{\partial y} dy$

$v(x, y + dy) = v(x, y) + \frac{\partial v(x, y)}{\partial y} dy$

During deformation, line  $PS$  is transformed into line  $P'S'$ . Since small deformations are assumed ( $\alpha \ll 1$ ,  $\cos \alpha \approx 1$ ), the lengths of  $P'S'$  and of its projection on  $y$  are approximately the same. So



$$|\overrightarrow{P'S'}| \approx \left( y + dy + v + \frac{\partial v}{\partial y} dy \right) - (y + v) = dy + \frac{\partial v}{\partial y} dy$$

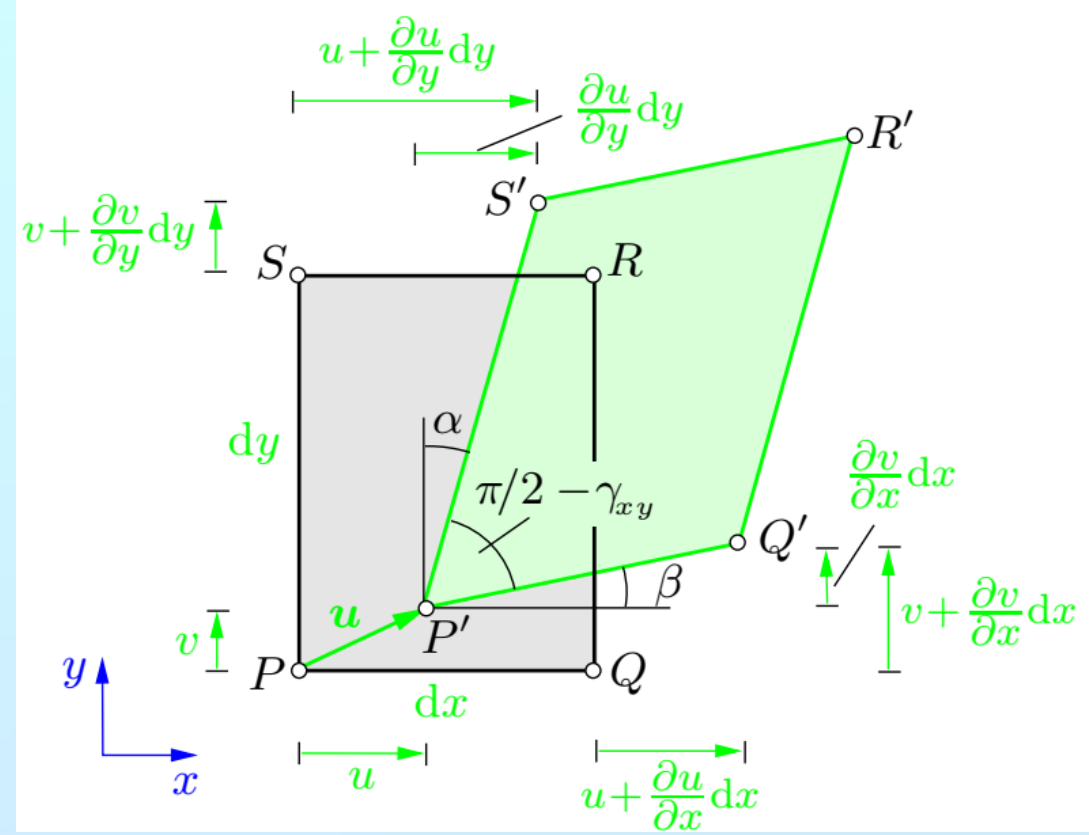
Then the *normal strain*  $\epsilon_y$  in the  $y$ -direction defined as the ratio of length increment to initial length, is

$$\epsilon_y = \frac{|\overrightarrow{P'S'}| - |\overrightarrow{PS}|}{|\overrightarrow{PS}|} = \frac{dy + \frac{\partial v}{\partial y} dy - dy}{dy} \Rightarrow \epsilon_y = \frac{\partial v}{\partial y}$$

For point  $Q$ :  $u(x + dx, y) = u(x, y) + \frac{\partial u(x, y)}{\partial x} dx$   
 $v(x + dx, y) = v(x, y) + \frac{\partial v(x, y)}{\partial x} dx$

For point  $S$ :  $u(x, y + dy) = u(x, y) + \frac{\partial u(x, y)}{\partial y} dy$   
 $v(x, y + dy) = v(x, y) + \frac{\partial v(x, y)}{\partial y} dy$

During deformation, initially right angle  $\widehat{QPS} = \frac{\pi}{2}$  is transformed into angle  $\widehat{Q'P'S'} = \frac{\pi}{2} - \gamma_{xy}$ .  
 For assumed small deformations ( $\alpha, \beta \ll 1$ ), the *shear strain (angular strain)* is given as:



$$\gamma_{xy} = \alpha + \beta \approx \tan \alpha + \tan \beta = \frac{u + \frac{\partial u}{\partial y} dy - u}{dy + \epsilon_y dy} + \frac{v + \frac{\partial v}{\partial x} dx - v}{dx + \epsilon_x dx} = \frac{\frac{\partial u}{\partial y}}{1 + \epsilon_y} + \frac{\frac{\partial v}{\partial x}}{1 + \epsilon_x}$$

Again for small deformation assumption ( $\epsilon_x, \epsilon_y \ll 1$ ), So the last expression is approximated to:

$$\gamma_{xy} \approx \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}$$

## Conclusion

Displacement of  $P(x, y)$  has two components:  
 $u(x, y)$  in  $x$  direction and  $v(x, y)$  in  $y$  direction.

These two displacements vary with point  $P$ , so they are functions of  $x$  and  $y$ .

Consequently they have the following four 1<sup>st</sup> order partial derivatives with respect to variables  $x$  &  $y$ :

$$\frac{\partial u}{\partial x}, \frac{\partial u}{\partial y}, \frac{\partial v}{\partial x} \text{ \& \ } \frac{\partial v}{\partial y}$$

Which are generally functions of the variables  $x$  &  $y$ .

They define the state of strain (deformation) at any point in the planar body by the three relations:

*x-normal strain*

$$\epsilon_x = \frac{\partial u}{\partial x}$$

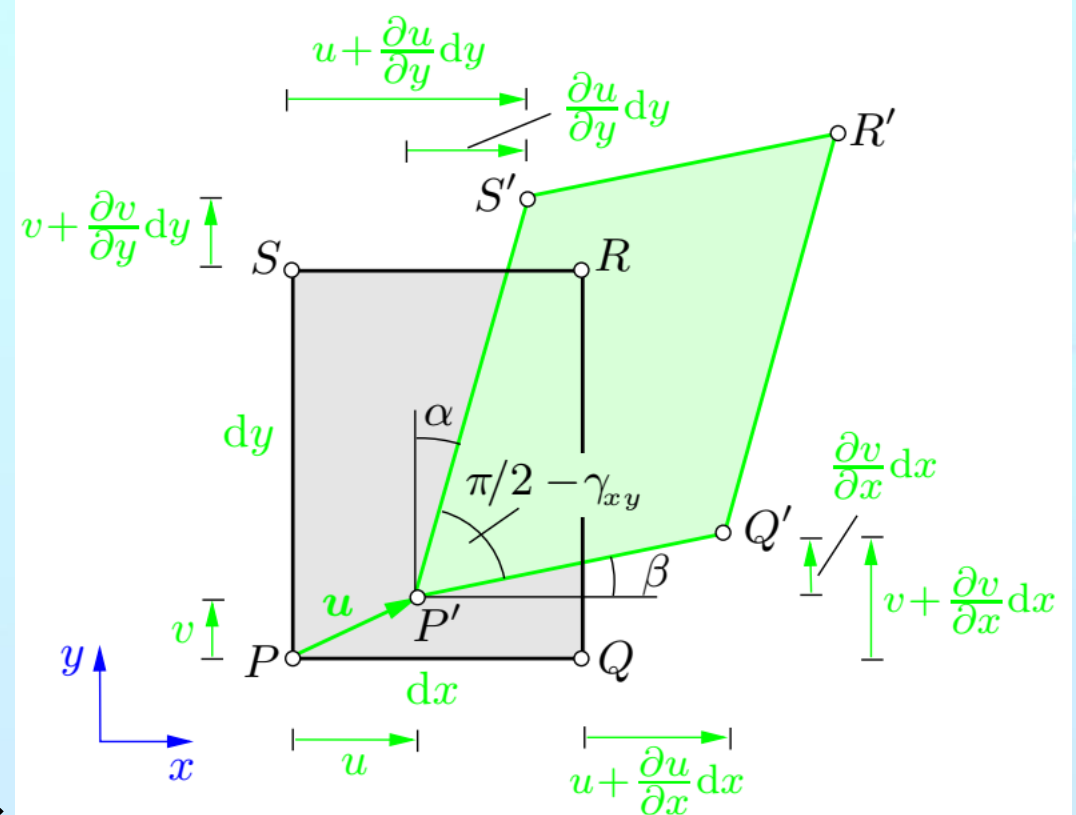
*shear strain*

$$\gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}$$

*y-normal strain*

$$\epsilon_y = \frac{\partial v}{\partial y}$$

These are the kinematic equations relating state of displacements to the state of strains.



## 3.2 Hooke's Law

The strains in a structural member depend on the external loading and therefore on the stresses.

Stresses and strains are connected by Hooke's law. In the uniaxial case it takes the form  $\sigma = E \varepsilon$

where  $E$  is Young's modulus. **Now Hooke's law will be formulated for the planar states of stress and strain.**

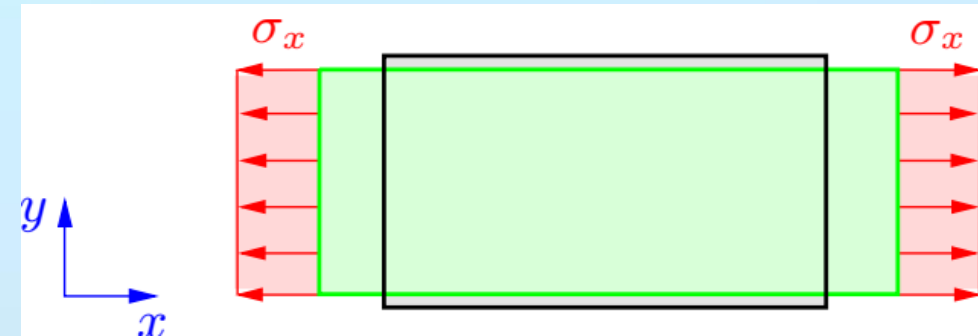
The formulation will be restricted to materials which are *homogeneous* and *isotropic*.

A *homogeneous* material has the same properties at each material point. For an *isotropic* material these properties are independent of the direction. An example of an anisotropic material is wood.

To derive Hooke's law for planar case, consider a rectangular domain which is cut out from a planar body and which is loaded only by the normal stress  $\sigma_x$ . Then, according to

Hooke's law in the uniaxial case:  $\sigma_x = E \varepsilon_x$  or  $\varepsilon_x = \frac{\sigma_x}{E}$ .

Experiments show that the tensile stress  $\sigma_x$  causes not only an increase of length but also a reduction of width of the rectangle. Thus, also a strain  $\varepsilon_y$  in  $y$ -direction appears. This phenomenon is called *lateral contraction* or *Poisson effect* (Simeon Denis Poisson, 1781–1840). The lateral strain  $\varepsilon_y$  is proportional to the axial strain  $\varepsilon_x$  and can be written as :  $\varepsilon_y = -\nu \varepsilon_x$ ,  $\nu$  is a material property called Poisson ratio.





*Poisson's ratio* is a material constant which is determined from experiments. Most metallic materials exhibit values about  $\nu \approx 0.3$ . Generally it can be shown that Poisson's ratio must be in the range of  $0 \leq \nu \leq 1/2$ .

As just discussed, the stress  $\sigma_x$  induces the strains  $\varepsilon_x = \frac{\sigma_x}{E}$  and  $\varepsilon_y = -\nu\varepsilon_x = \frac{-\nu\sigma_x}{E}$ .

Analogously, a stress  $\sigma_y$  induces the strains  $\varepsilon_y = \frac{\sigma_y}{E}$  and  $\varepsilon_x = -\nu\varepsilon_y = \frac{-\nu\sigma_y}{E}$ .

When both stresses,  $\sigma_x$  as well as  $\sigma_y$ , are acting, the total strains are obtained by superposition:

$$\varepsilon_x = \frac{1}{E} (\sigma_x - \nu\sigma_y) \quad \text{and} \quad \varepsilon_y = \frac{1}{E} (\sigma_y - \nu\sigma_x)$$

If the domain which cut out from a planar body is loaded solely by shear stresses  $\tau_{xy}$  (pure shear), a linear relationship between the angle change  $\gamma_{xy}$  and  $\tau_{xy}$  is experimentally observed:

$$\tau_{xy} = G\gamma_{xy}$$

The constant  $G$  is called *shear modulus*. It is a material parameter which can be experimentally determined in a shear test or a torsion test. The shear modulus  $G$  has the same dimension as Young's modulus  $E$ , i.e. force/area, and it is usually expressed in MPa.

It can be shown that there exist only *two independent* material constants for isotropic, linear elastic materials. The following relationship holds between the three constants  $E$ ,  $G$  and  $\nu$ :

$$G = \frac{E}{2(1 + \nu)}$$

**Conclusion** If the state of stress in a planar homogeneous, isotropic body is described by

$$\sigma_x(x, y), \sigma_y(x, y) \text{ and } \tau_{xy}(x, y)$$

Then the state of strain described by

$$\varepsilon_x(x, y), \varepsilon_y(x, y) \text{ and } \gamma_{xy}(x, y)$$

Is related at every point to the state of stress via the materials properties  $E, G, \nu$ , by Hook's law:

$$\varepsilon_x = \frac{1}{E} (\sigma_x - \nu\sigma_y)$$

$$\varepsilon_y = \frac{1}{E} (\sigma_y - \nu\sigma_x)$$

$$\gamma_{xy} = \frac{1}{G} \tau_{xy} = \frac{2(1 + \nu)}{E} \tau_{xy}$$

Hook's law can also be written in the inverse form

$$\sigma_x = \frac{E}{1-\nu^2} (\varepsilon_x + \nu\varepsilon_y)$$

$$\sigma_y = \frac{E}{1-\nu^2} (\nu\varepsilon_x + \varepsilon_y)$$

$$\tau_{xy} = G\gamma_{xy} = \frac{E}{2(1+\nu)} \gamma_{xy}$$