

# Cyclic Loading

Engineering systems are often subjected to *cyclic loading*.  
Automobiles & machines (millions of cycles).

Electronic components (fluctuating temperature).

Material degradation with time under *cyclic loading* is known as *fatigue* (التعب).

Fatigue can cause materials to fail at stress levels well below their uniaxial strength. In many steels, the *fatigue strength* is less than half of the *ultimate strength*.

A few definitions are useful to describe the nature of cyclic loading.

$$\sigma_m = \frac{\sigma_{max} + \sigma_{min}}{2}$$

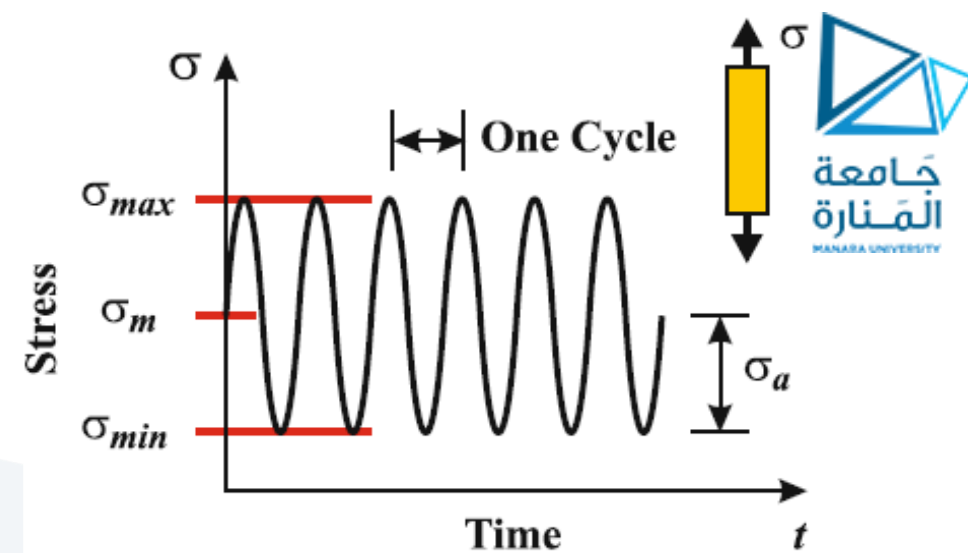
Mean Stress الإجهاد المتوسط

$$\sigma_a = \frac{\sigma_{max} - \sigma_{min}}{2}$$

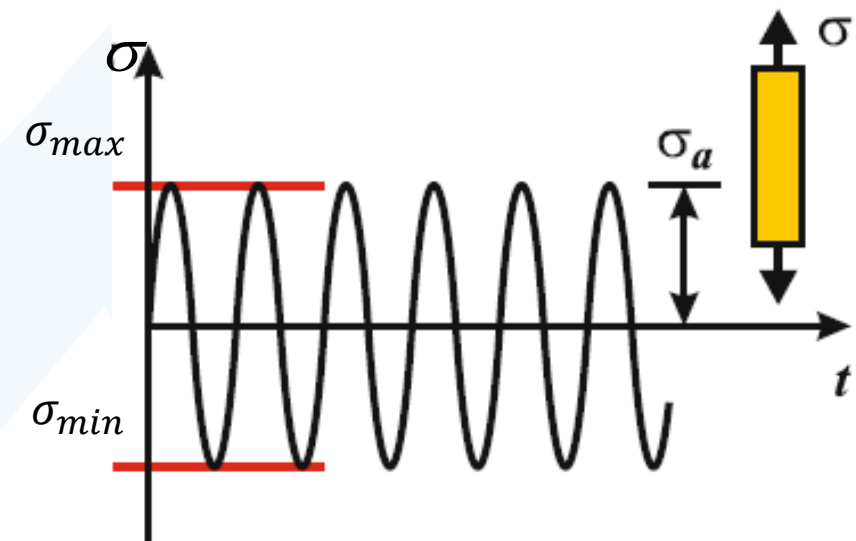
Stress amplitude مطال الإجهاد

$$R = \frac{\sigma_{min}}{\sigma_{max}}$$

R-ratio



Typical  $\sigma - t$  graph for cyclic loading.



$\sigma - t$  graph for standard fatigue test:  
 $\sigma_a = \sigma_{max} = |\sigma_{min}|, \sigma_m = 0, R = -1.0.$

When the stress amplitude is plotted against the number of cycles to failure on a log–log set of axes (or on a semi-log set of axes), the graph is known as the ***S–N curve***

The ***fatigue strength*** is the stress amplitude  $\sigma_a = \sigma_{\max}$  corresponding to a specified number of cycles to failure.

When the data are plotted, the relationship between stress amplitude  $\sigma_a$  and cycles to failure  $N_f$  can be expressed in the following form:

$$\sigma_a = S'_f (N_f)^b$$

Where  $S'_f$  &  $b$  depend on the material, & are determined from a **best-fit line** of the material's ***S–N curve*** since:

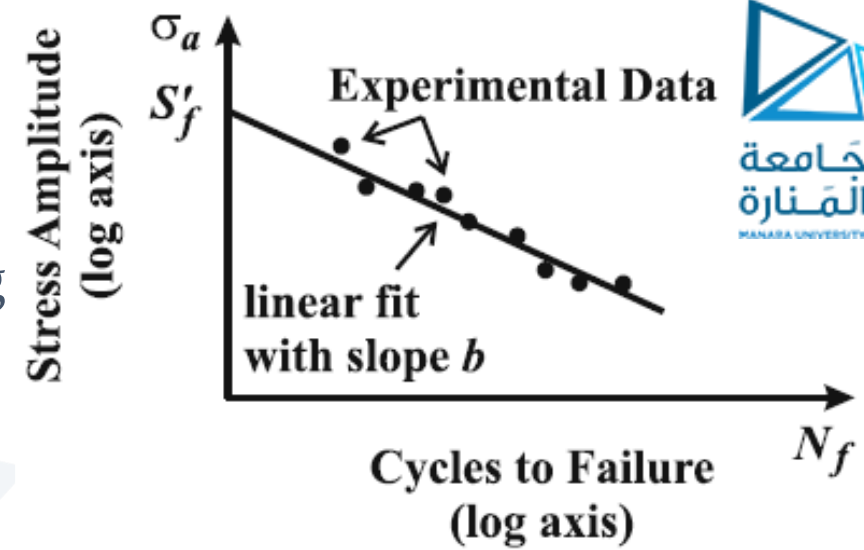
$$\log(\sigma_a) = \log(S'_f) + b \log(N_f)$$

Quantity  $S'_f$  corresponds to the curve fit's intercept at  $N_f = 1$  cycle,  $b$  is the slope of the curve in the log–log axes. Values for aluminum & steel are given

### ***Ex. 1 Fatigue Strength of Structural Aluminum & Steel***

**Given:** A tensile bar is subjected to cyclic loading, with zero mean stress

**Required:** Determine the fatigue strength,  $\sigma_a = \sigma_{\max}$ , (a) for an aluminum bar (Al 6061-T6) at  $10^7$  cycles and (b) for a steel bar (A36) at  $10^6$  cycles.



Property	Al 6061-T6	Steel A36
$S_y$	240 MPa	250 MPa
$S_u$	314 MPa	540 MPa
$S'_f$	505 MPa	1035 MPa
$b$	-0.082	-0.11
$E$	70 GPa	200 GPa

**Solution:** For aluminum, the fatigue strength corresponds to  $10^7$  cycles. From *Table*  $S'_f = 505 \text{ MPa}$  &  $b = -0.082$ .

$$\sigma_{a,al} = S'_f(N_f)^b = (505 \text{ MPa})(10^7)^{-0.082} = 135 \text{ MPa. (56\% } S_y, 43\% S_u)$$

For steel, the fatigue strength corresponds to  $10^6$  cycles. From *Table*  $S'_f = 1035 \text{ MPa}$  &  $b = -0.11$ .

$$\sigma_{a,st} = S'_f(N_f)^b = (1035 \text{ MPa})(10^6)^{-0.11} = 226 \text{ MPa. (90\% } S_y, 42\% S_u)$$

These values are also known as *fatigue limit*  $S_{FL}$ , as discussed below.

The fatigue limit for many steels is typically on the order of 35 - 50% of  $S_u$ .

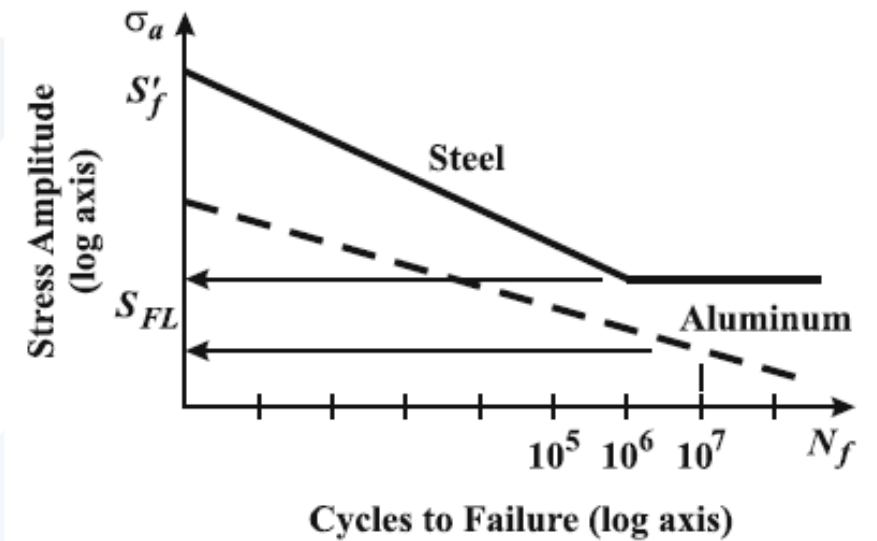
## Fatigue Limit

The ***S–N curve*** for most steels doesn't decrease to zero (*Fig.*). Below a certain value of  $\sigma_a$ , steel has essentially an infinite fatigue life.

The stress amplitude below which fatigue failure does not occur is the ***fatigue limit***  $S_{FL}$ , also known as the endurance limit **حد التحمل**.

Type equation here. For steels, the ***fatigue limit***  $S_{FL}$  typically corresponds to a fatigue life of about  $10^6$  cycles. The S–N curve exhibits no further reduction and becomes horizontal.

Aluminum does not exhibit such limiting behavior. No matter how small the stress amplitude  $\sigma_a$ , aluminum will eventually fail by fatigue. Accordingly, when designing with aluminum, the fatigue limit  $S_{FL}$  is generally taken to be the stress amplitude for  $N_f = 10^7$  cycles.



# Effect of Mean Stress on Fatigue Strength

In general, cyclic stresses are applied with a non-zero mean stress  $\sigma_m$ .

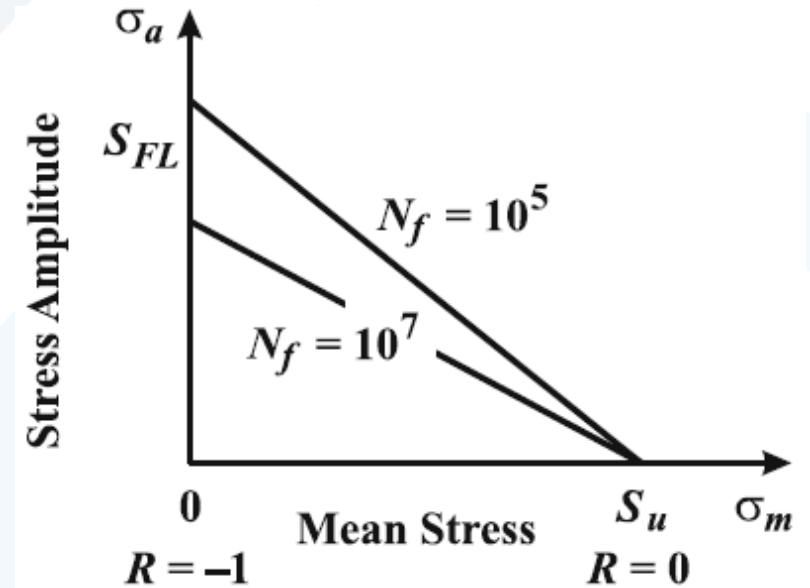
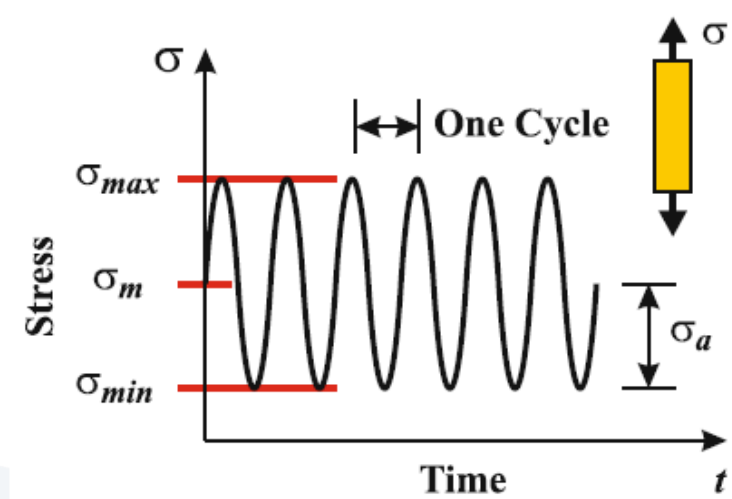
When this is the case, the fatigue strength for a given number of cycles to failure  $N_f$  is determined using the **Goodman Diagram**

$$\frac{\sigma_a}{S_{FL}} + \frac{\sigma_m}{S_u} = 1$$

The new fatigue strength  $\sigma_a$  – the amplitude of the cyclic loading with mean stress  $\sigma_m$  for the specified cycles to failure  $N_f$  – is then:

$$\sigma_a = S_{FL} \left( 1 - \frac{\sigma_m}{S_u} \right)$$

When  $\sigma_m = S_u$ , the material breaks into two upon first loading; there can be no alternative stress  $\sigma_a$  ( $\sigma_a = 0$ ). If the mean stress is zero, then the stress amplitude is the *fatigue limit*  $S_{FL}$  for the specified number of cycles to failure.



## Ex. 2 Fatigue Strength With Non-Zero Mean Stress

**Given:** A tensile bar is subjected to cyclic loading with a mean stress of  $\sigma_m = 100$  MPa.

**Required:** With the mean stress applied, determine the fatigue strength (a) for an Al 6061-T6 bar at  $10^7$  cycles and (b) for an A36 steel bar at  $10^6$  cycles.

**Solution:**

$$\sigma_{a,al} = S_{FL} \left( 1 - \frac{\sigma_m}{S_u} \right) = 135 \left( 1 - \frac{100}{314} \right) = 91 \text{ MPa.}$$

$$\sigma_{a,st} = S_{FL} \left( 1 - \frac{\sigma_m}{S_u} \right) = 226 \left( 1 - \frac{100}{540} \right) = 184 \text{ MPa.}$$

